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development at Honeywell ICBM reentry vehicle navigation system

Boutelle, J. Kau, S.P. Marino, C.J., Jr.

Symposium, IEEE 1998 This paper appears in: Position Location and Navigation Div. of Space & Strategic Studies, Honeywell Inc., USA;

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upon the need to identically match the physical properties Space Company) contracted Honeywell to develop an and Space Division (now the Lockheed Martin Missiles and introduced on this program. In 1991, the Lockheed Missiles reentry vehicles. The Reentry Inertial Measurement Unit instrumentation package for the US Navy Trident system the laser gyro, with advanced radiation hardening, was development was followed by the USAF Advanced Inertial the needed reentry flight navigation accuracy. The DINS excellent linearity of the Bell XI accelerometer, provided degree's rotation experienced in the coast phase of flight. strapped down technology, demonstrated the ability of tile navigation for the US Air Force Advanced Maneuvering the Dormant Inertial Navigation System (DINS), provided spinning coast phase, and high acceleration reentry phase, Measurement Unit (AIMU) project. The next generation of development of the Ring Laser Gyro. The original product, of a reentry vehicle (RV) flight was enabled by the Ring Laser Gyro to accurately measure the millions of Reentry Vehicle (AMaRV). The DINS, which employs Maintaining an accurate navigation solution through the (RIMU) program presented unique design constraints based The laser gyro's minimal sensitivity to acceleration, and the launched by ICBMs was initiated at Honeywell in 1977. Development of navigation systems for reentry vehicles

experience provides the basis for a new generation of vehicle navigation as conducted by Honeywell Space and paper presents the history of, and lessons learned during, performance navigation systems for improved ballistic missile Strategic Systems from 1977 through 1997. This the development of hardware and software for reentry software tools to enable system and sensor configuration strapped down technology. We have also developed the advancements in the real-time navigation software for DINS program, has matured on the RIMU program. This Analysis software tool development, which started on the design trades for new missions. Additionally, our Post Flight and RIMU projects, Honeywell has achieved significant gyros and gimbaled platform technology. The Honeywell of the predecessor IMU that was based upon spinning mass RIMU has performed reliably and accurately on several Trident flights. Through experience on the AMaRV, AIMU,

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